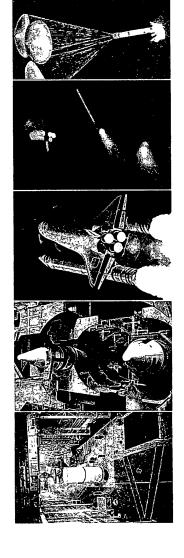
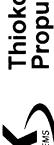
Automation of NDR Metal Components S Y S Y S C O

Presented by:

John Hartman, ATK Thiokol Propulsion Mark Kirby, Westinghouse AMDATA









Automation of NDE on RSRM components

Acknowledgements

The authors would like to acknowledge the following people and organizations for their contributions to this effort

Mike Suits Craig Bryson Scott Teunis Dave Kay

Marshall Space Flight Center, NASA Marshall Space Flight Center, NASA Progressive Technologies, Inc.

ATK Thiokol Propulsion



Automation of NDE on RSRM components

Past NDE on RSRM metal components

Why change?

Advantages/Improvements

Elements of Automated Inductive Inspection System

Current Status



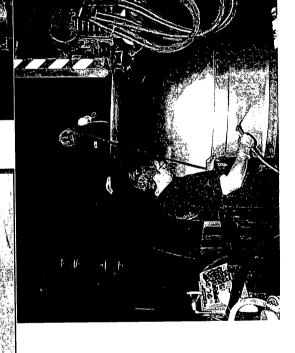
Past NDE on RSRM metal components

- Historically, nearly all NDE has been visually based
- Magnetic particle inspection of steel components
- Liquid penetrant of aluminum components
- Semi-automated and manual eddy current inspections of select holes and joints
- discontinuities in both new and refurbished metal Goal of inspections: detect surface cracks or components

Past NDE on RSRM metal components

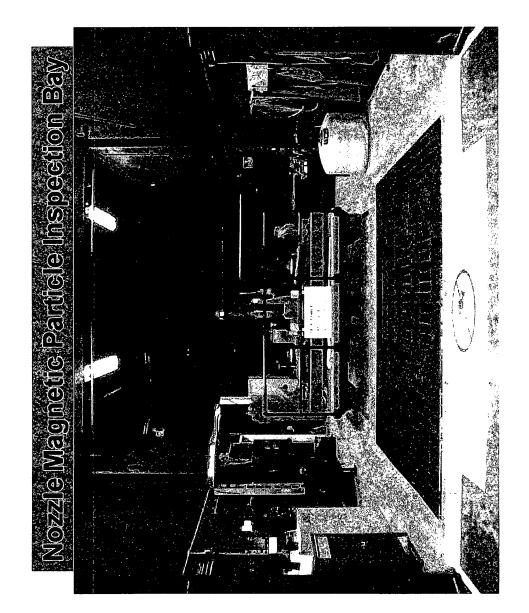
وععه الالعواسطاه العلياط العلقاس العلامة الالعقائدة





ATTK ALLIANT TECHSYSTEMS

Past NDE on RSRM metal components



ATTK ALLIANT TECHSYSTEMS

Past NDE on RSRM metal components





Past NDE on RSRM metal components

大工

CHANGE?



Why Change?

Biggest Reason: Increase Reliability

Reliability: Assurance that critical sized flaws will not go undetected

Remove the strong dependency on operator skill and attentiveness



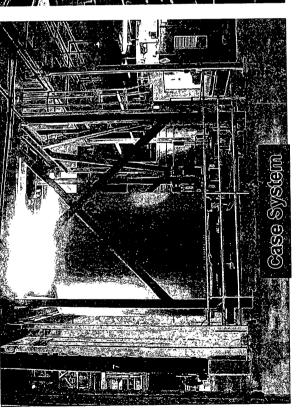
Why Change?

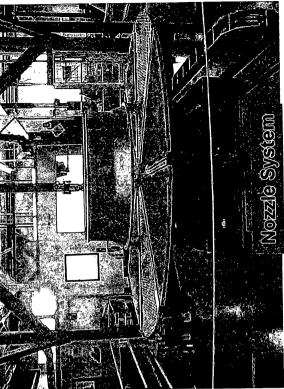
- Safe flight is certified by Proof Test, NDE, or both
- inspections do not reliably screen for Critical Flaws in ALL Based on in-house POD study, current magnetic particle regions
- Minimum Detectable Flaw Size (90/95) > Critical Flaw Size
- These areas must be certified by proof and/or eddy current



Why Change?

- Solution:
- (AIIS) with the capability of reliably detecting critical sized Implement an Automated Inductive Inspection System surface flaws







Control & Repeatability

- Inspections, data acquisition & analysis are programmed
- Same inspection & analysis each time
- Detection sensitivity is calibrated
- Pressure is no longer on the operator & his attentiveness



Better detection capability

- Overall, AIIS will find smaller flaws more reliably (at a 90%POD/95% CL)
- Current sensitivity levels are set at 0.1" and 0.25" long cracks (2:1 length-to-depth ratio)



Data Storage

- All data and images are stored
- determining the nature of flaws (manufacturing vs. service) This can be useful in developing history for a part and for



Reduction of waste streams

- No solvents or chemicals to dispose of
- Reduction or elimination of acid etch
- Also extends lifetime of aluminum parts

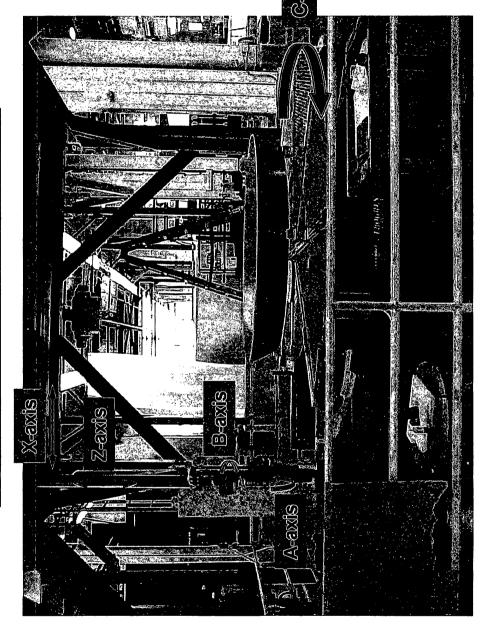


Potential reduction of process time for Case Hardware and Aluminum Nozzle Hardware

Possible elimination of acid etch process

Possible elimination of glass bead

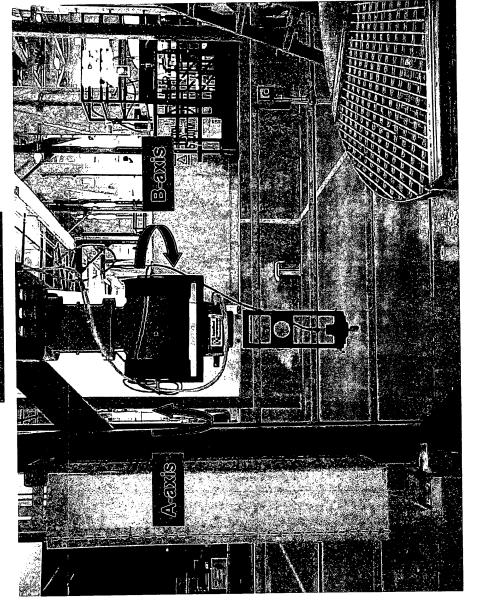
S Independent exes 7 exes when inspecting various holes

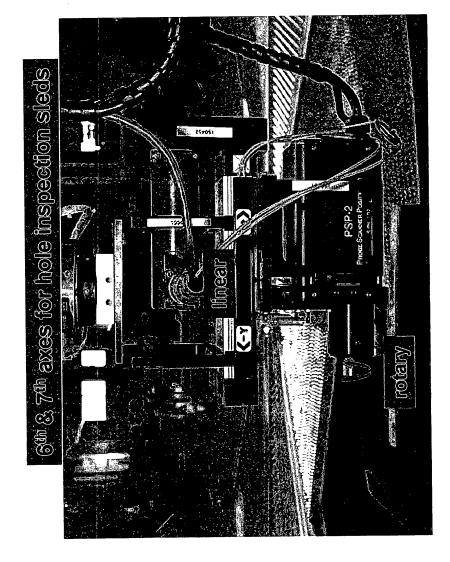


ALLIANT TECHSYSTEMS

Elements of AllS

Glose up of wrist





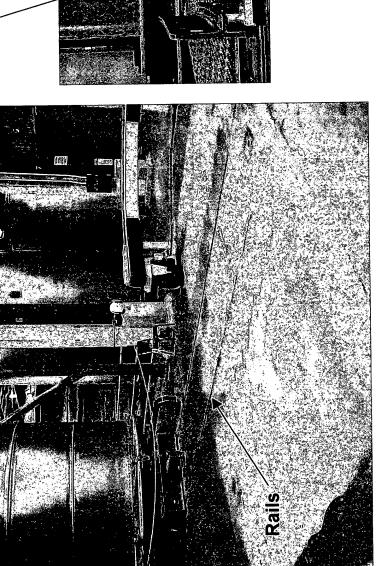


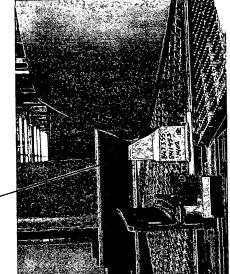


Elements of AIIS

Turntable is on rails, extends into high bay for loading and offloading of components

Components are mounted on adjustable chocks

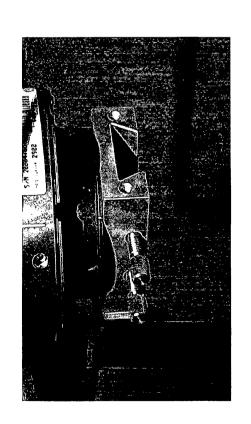


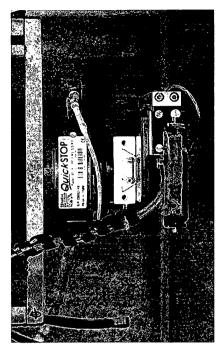




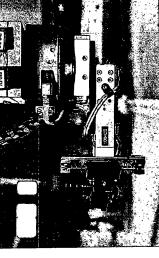
Elements of AIIS

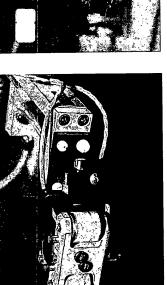
- Each component incorporates multiple "probe sleds" used to inspect different part geometries
- All sleds slide onto wrist using dovetail joint

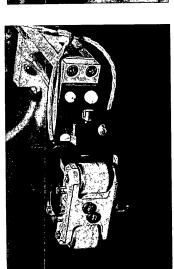




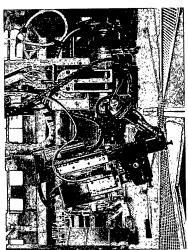


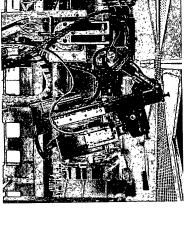














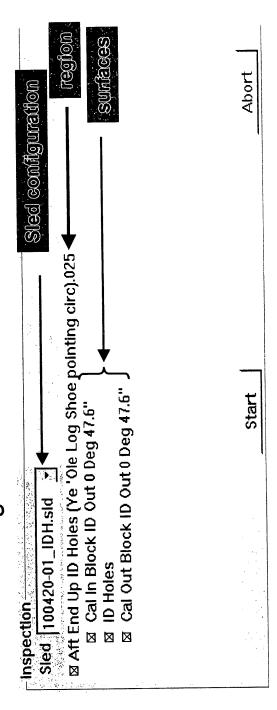
Elements of AIIS

- Operator interface: Part Inspection Program (PIP)
- Operator selects part; then probe sled

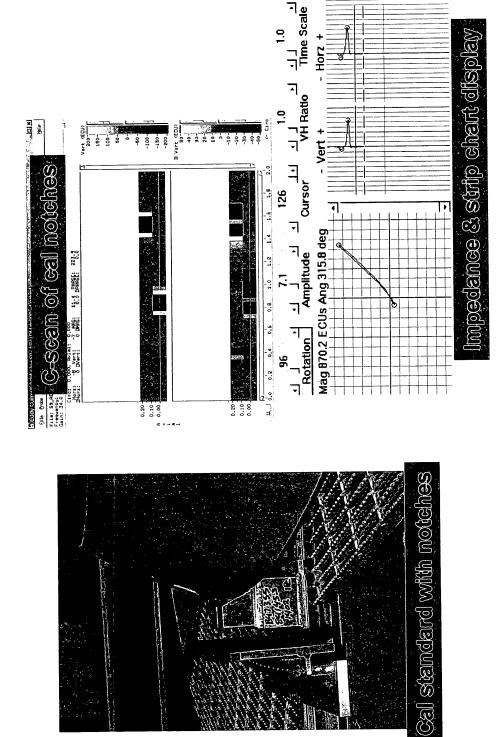
Sled 100420-01_IDH.sld	brit Alad
In Block ID Out 0 Deg 47.6" Fout Block ID Out 0 Deg 47.6" Start	
Out Block ID Out 0 Deg 47.6"	
Start Start	
Start	
Start Start	1 1
SIS	1
2. 4. 5. 4. 10. 5. 4. 4. 5. 10. 10. 10. 10. 10. 10. 10. 10. 10. 10	
Cal Out Block ID Out to beg 47:0	ন =
Art End Up ID Memorane Section Cal In Block ID Out 0 Deg 47.6"	1
ID Membrane Cal Out Block ID Out 0 Deg 47.6"	
Aft End Up OD Membrane Section	7 0
Results	
Aft End Up ID Holes (Ye 'Ole Log Shoe pointing circ).025 Cal In Block ID Out 0 Deg 47.5"	4 □
ID Holes Cal Out Block ID Out 0 Deg 47.6"	an makaneng pi meme
Aft End (Ye'Ole Log Shoe) 0.75 Cal in Block ID Out 0 Deg 47.6"	ন



- All regions and surfaces inspected with the sled are selected, and the AIIS inspects those surfaces
- Each "region" starts & ends with a calibration scan



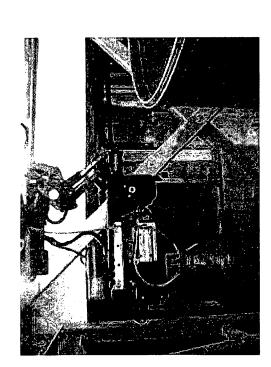


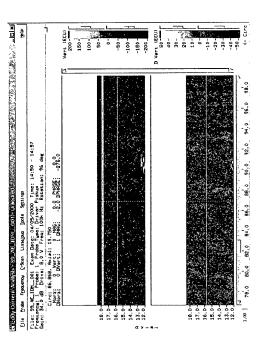




Elements of AIIS

Scan surface of interest

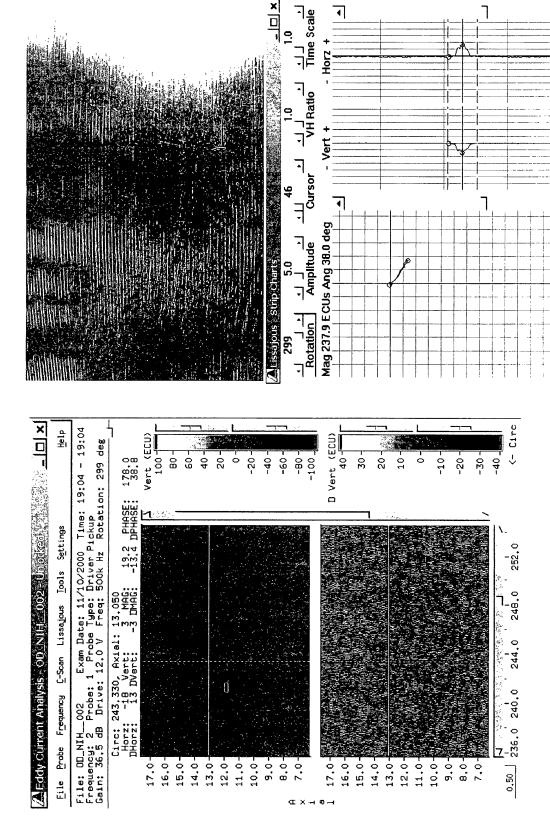




P

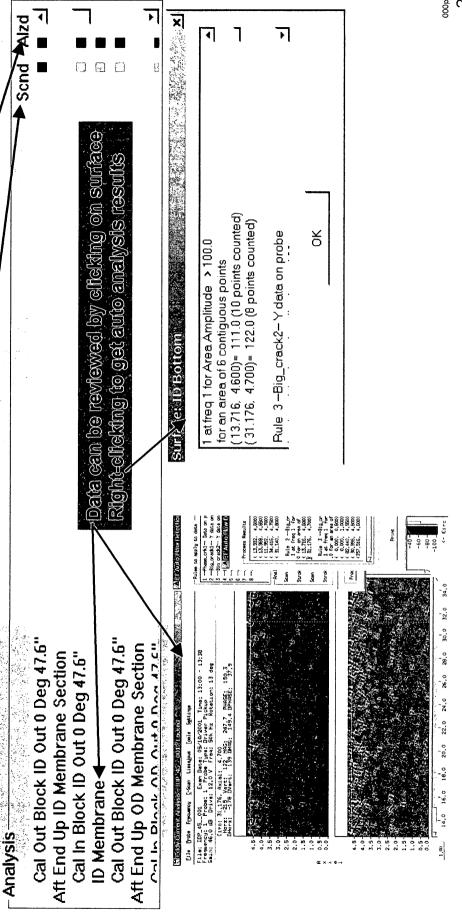


Elements of AIIS





PIP keeps track of data acquired and analyzed

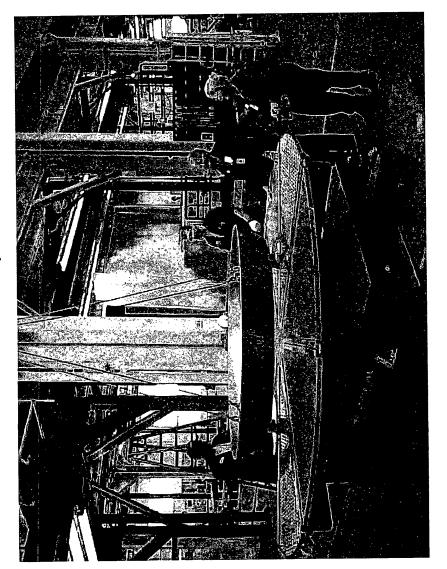




- After all data is acquired, analysis results can be printed out
- All auto analysis findings are reviewed for final evaluation
- C-scan images are also independently reviewed by the operator in a pseudo real-time mode



Part is off-loaded and next part is loaded on





Current Status

- AllS is now inspecting flight hardware concurrently with the certified NDE process (MT, PT, ET, UT)
- Once all qualification paperwork and approvals are obtained, the majority of the visual and manual inspections (MT, PT, ET) will be deleted
- Select areas that tend to more frequently have cracks will continue to be double inspected (AIIS & MT)



Concluding Statements

- inspect RSRM (Space Shuttle) metal components designed and built, and is being implemented to An automated eddy current system has been
- inspection reliability, as well as other benefits such as data storage, chemical waste reduction and reduction The system provides a significant increase in in overall process time